

# COE CST Sixth Annual Technical Meeting

## Task 323 Structural Health Monitoring Framework

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### Contributors

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# Task 323 SHM Framework

- Task 323 focuses on SHM architecture and guidelines for integrating SHM with spaceflight recorder (aka “black box”).
- Investigation of thermal and radiation fatigue of smart structures for assessment of RVL’s condition during flight.
- Prepare hardware for evaluation of space effects on structural condition and sensor system.

# Motivation

## On-orbit Monitoring

- 1) Component identification and performance assessment.
- 2) Elements of mission and space weather monitoring.

## Re-entry Monitoring

- 1) Structural temperature and strain profiles.
- 2) Material degradation/breakup monitoring.

**Certified for flight / re-flight?**

## Monitoring During Launch

- 1) Monitoring launch environment.
- 2) Loads assessment during launch.

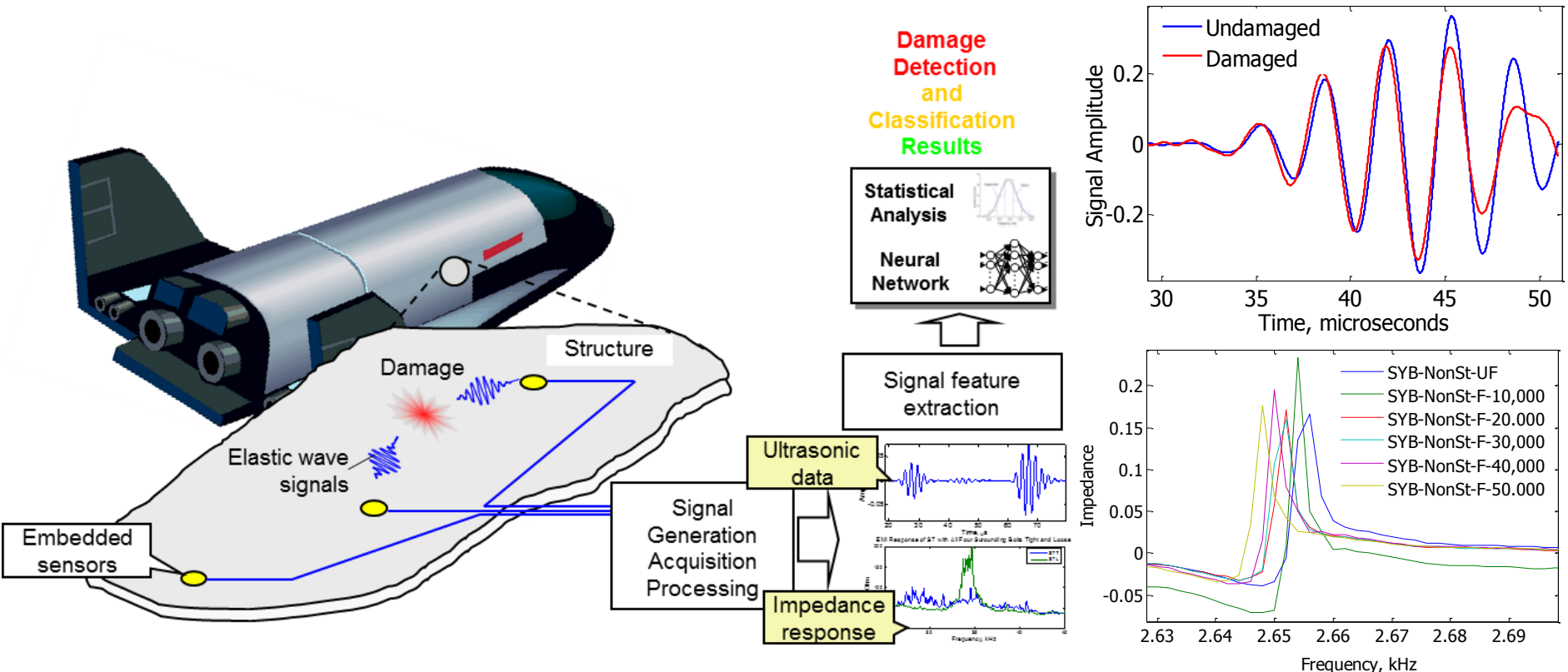
## Pre-launch Diagnosis

- 1) Assessment of structural integrity.
- 2) Assessment of critical interfaces and joints.
- 3) Remaining life prediction via SHM data/FEA correlation.

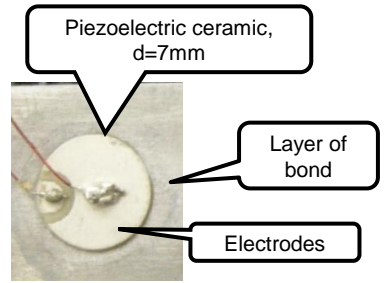
## SHM Modalities

Passive Monitoring During Flight  
+  
Active Monitoring on the Ground

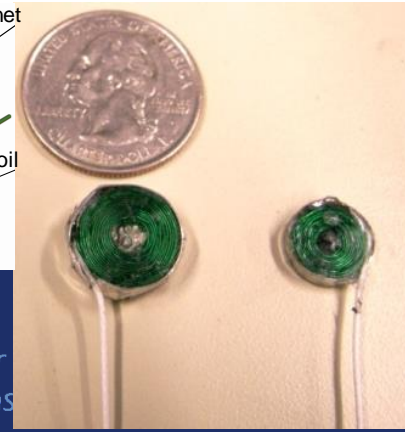
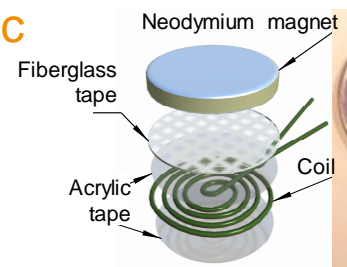
# Structural Health Monitoring



Thin  
Piezoelectric  
Sensors  
7mm x0.2mm

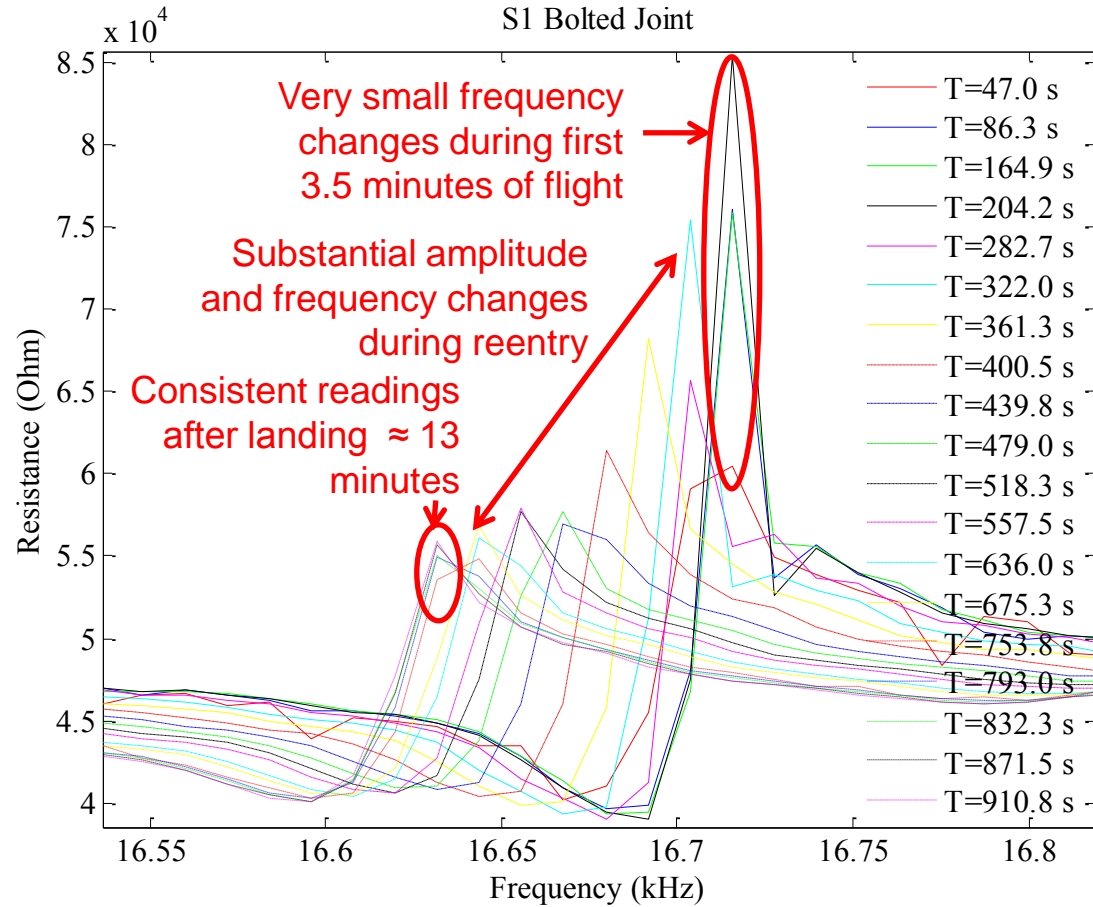
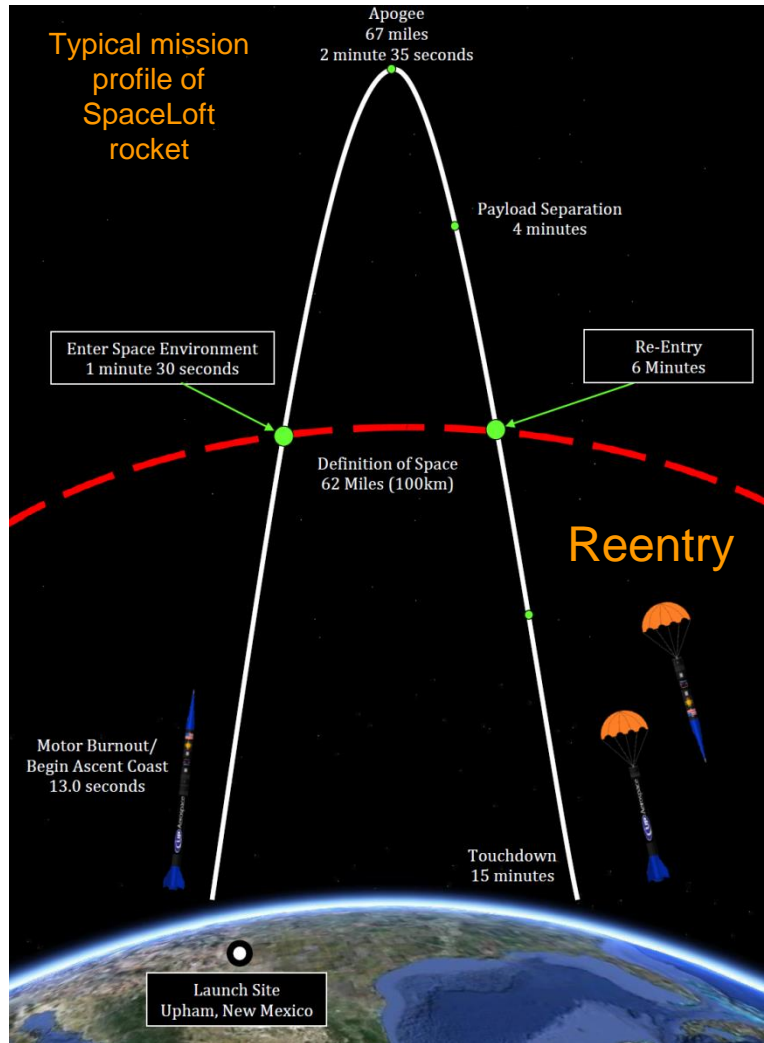


Magneto-elastic  
Active Sensor  
(MEAS)



# Suborbital flight data

## Electro-mechanical impedance spectra



# Extreme Space Environment

In space, structure will be exposed to

- Extreme temperatures
- Vacuum
- Radiation
- Atomic oxygen
- Micro-gravity
- Micro-meteroids and debris

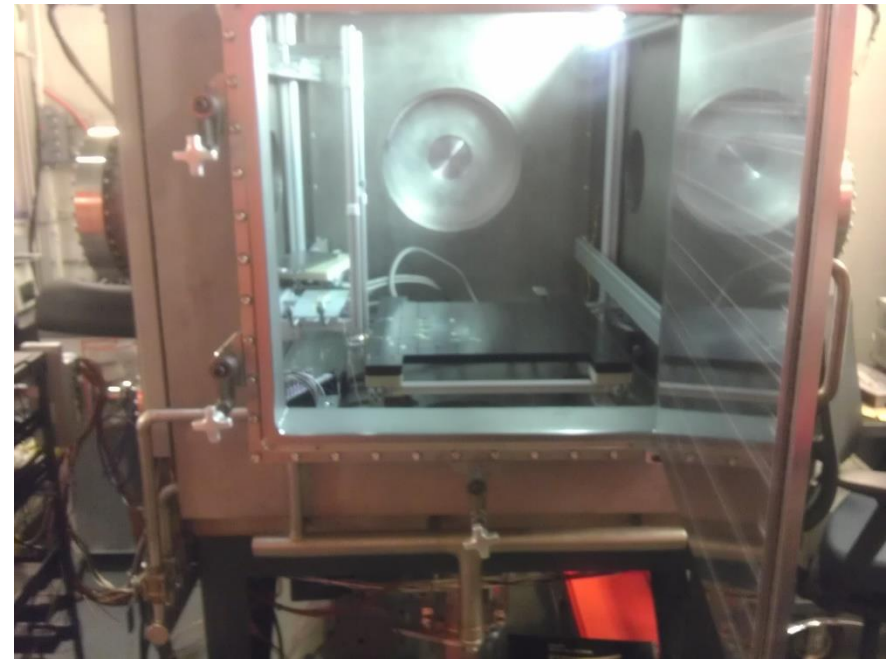
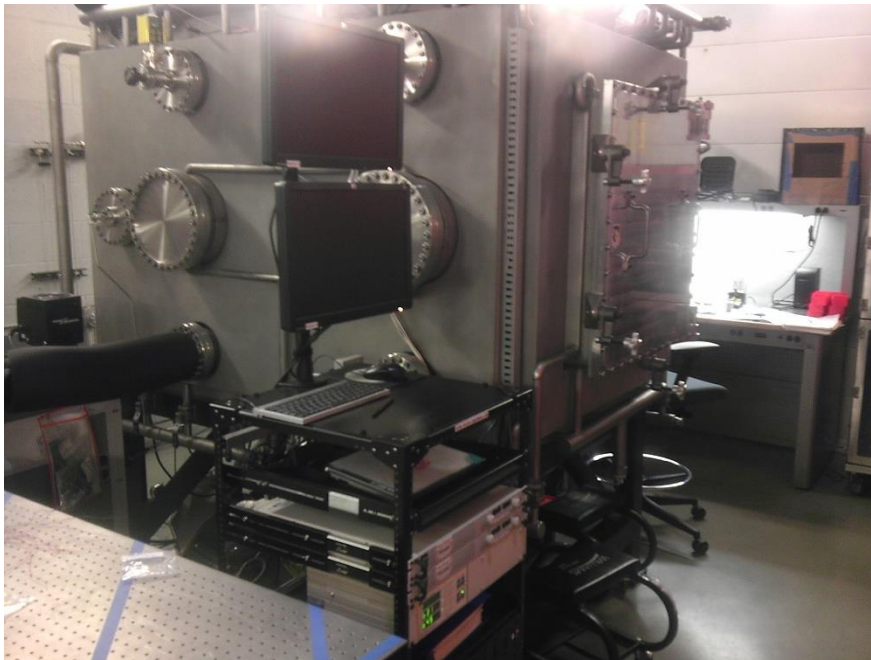
On Earth, aspects of space can be emulated

- Extreme temperatures and vacuum at AFRL
- Gamma radiation at WSMR

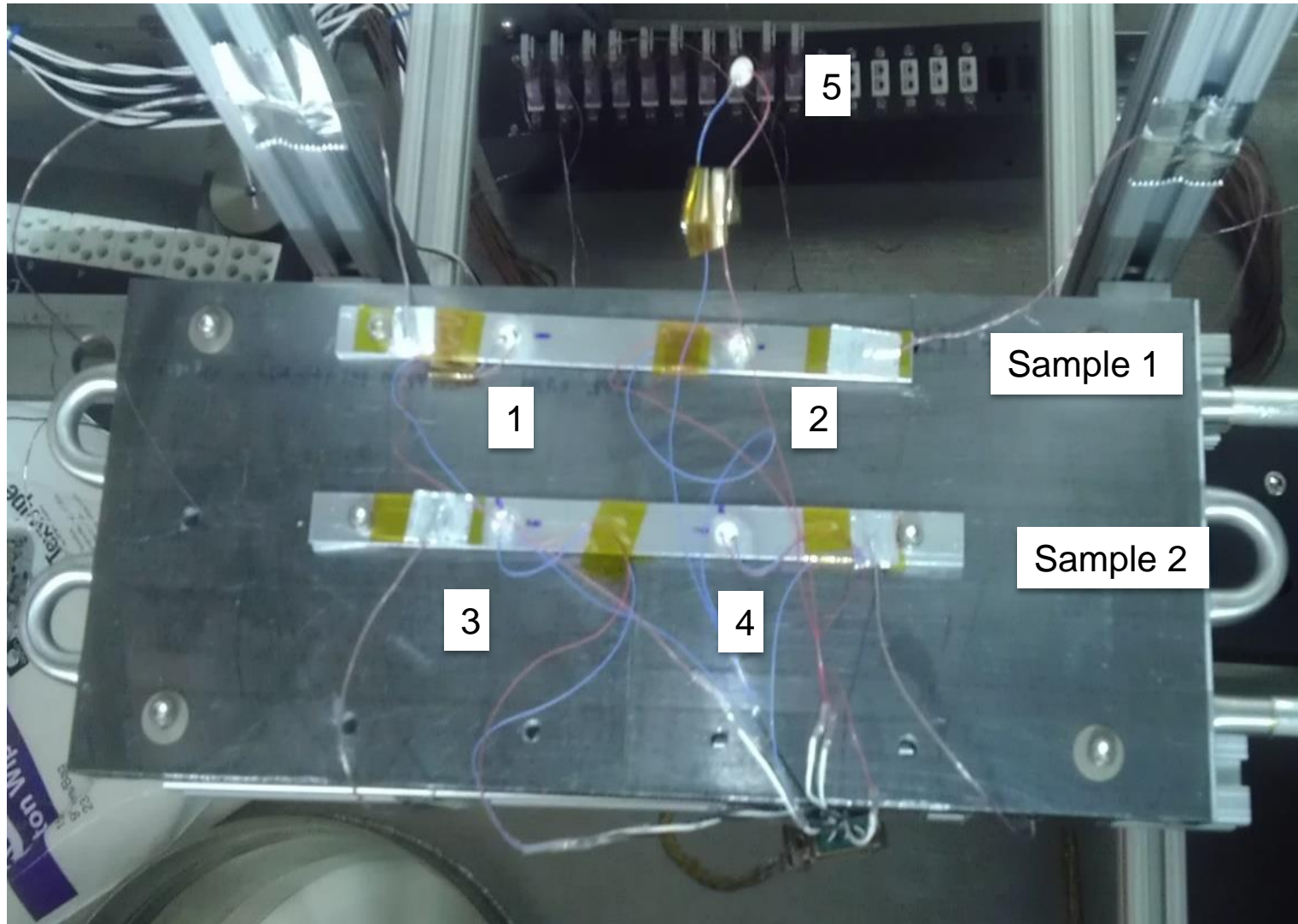
# Temperature and Vacuum Effects

## AFRL Vacuum Thermal Chamber

- Chamber Pressure  $2 \times 10^{-6}$  Torr
- FTS RC2111 Recirculating Chiller



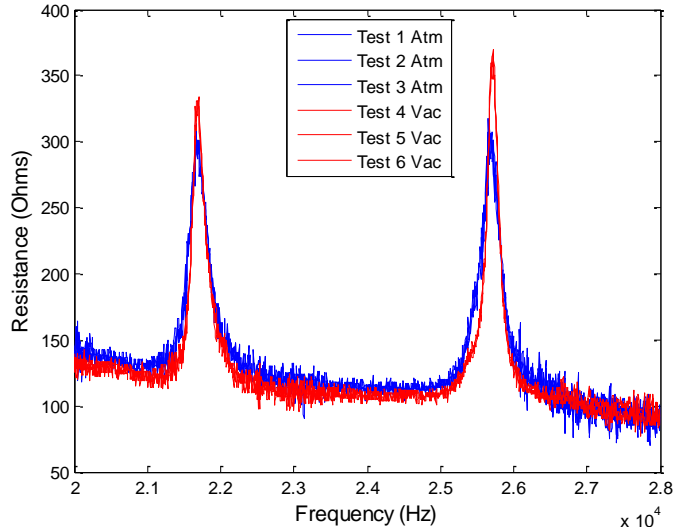
# Temperature and Vacuum Effects



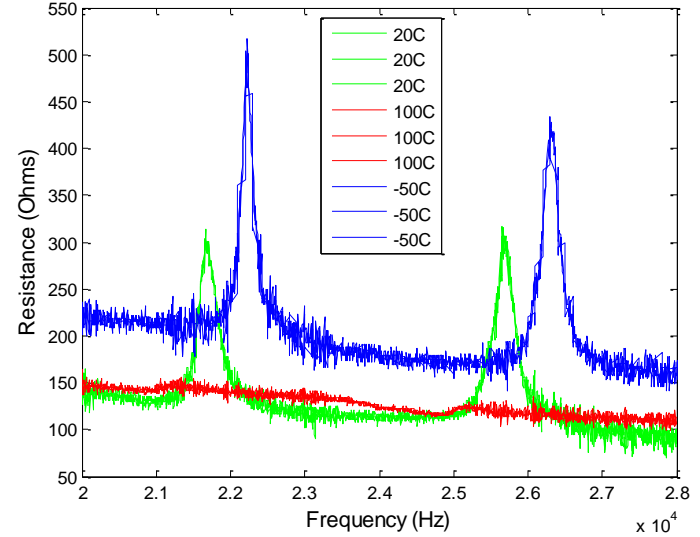


# Temperature and Vacuum Effects

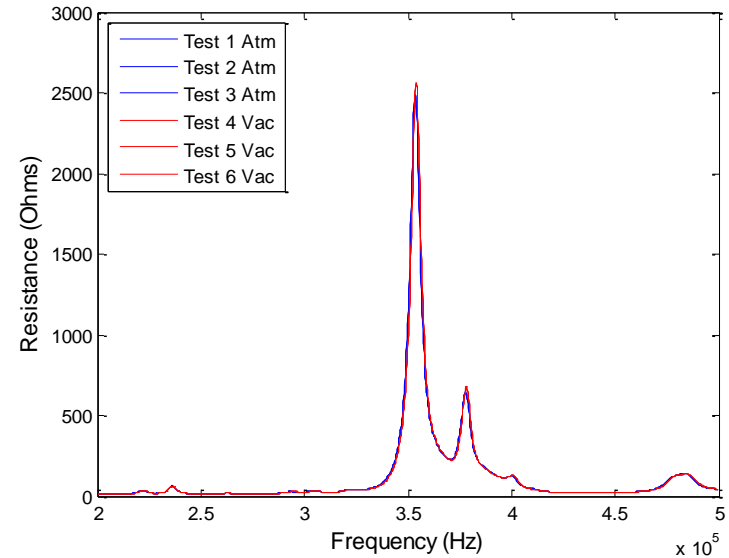
Sensor 1 @ 20C



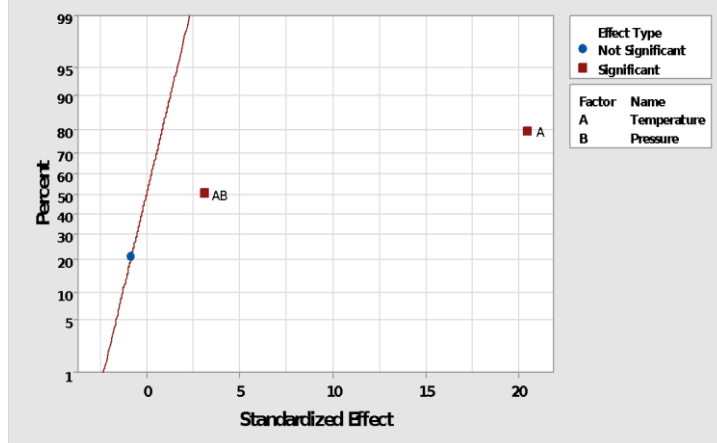
Sensor 1 @ Atmospheric Pressure



Sensor 5 @ 20C



Normal Plot of the Standardized Effects  
(response is Amplitude,  $\alpha = 0.05$ )



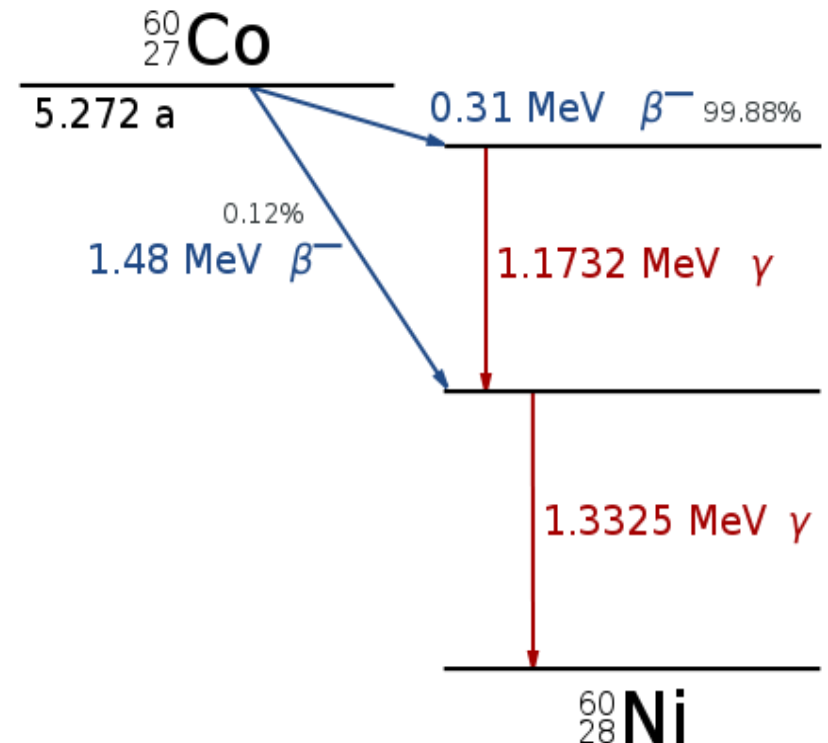
# Investigation of Radiation Effects

## Radiation in Space

- Background galactic cosmic radiation
- Solar event radiation
- Radiation from particles trapped in the Van Allen Belts

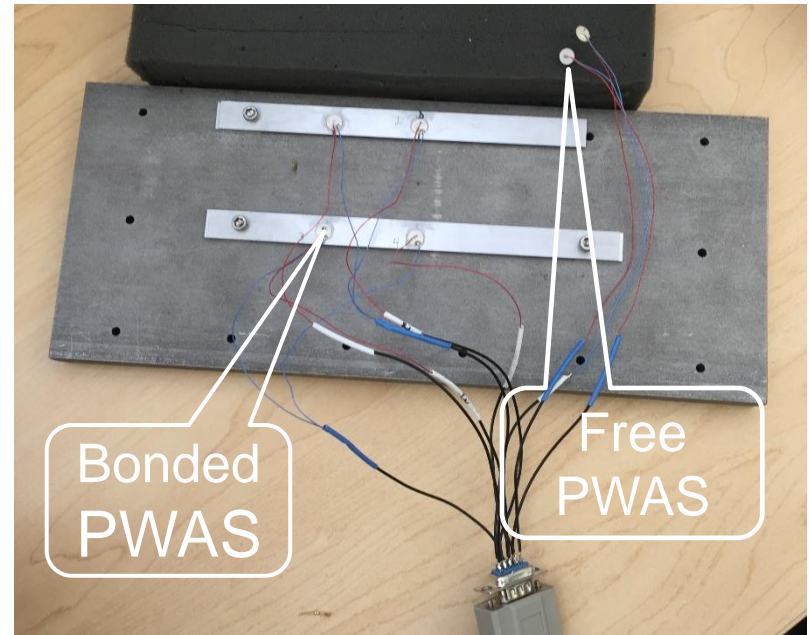
## Gamma Radiation

- Suitable for emulating space environment (ESA)

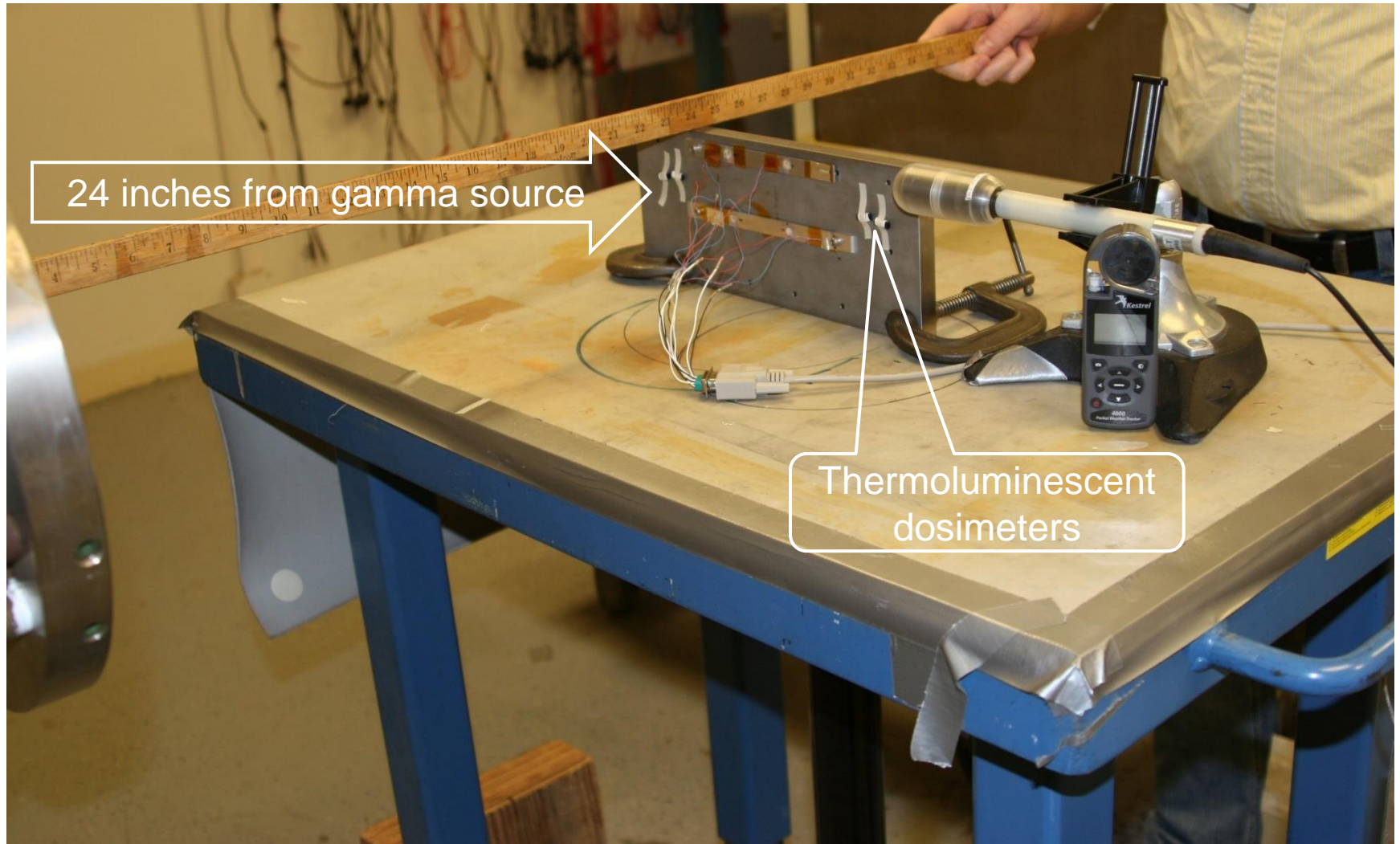


# White Sands Gamma Facility Test

- Cobalt 60 gamma radiation
- PZT sensors bonded to 6061-T6 aluminum beams
- Frequency sweeps with impedance analyzer
- Data collection and analysis



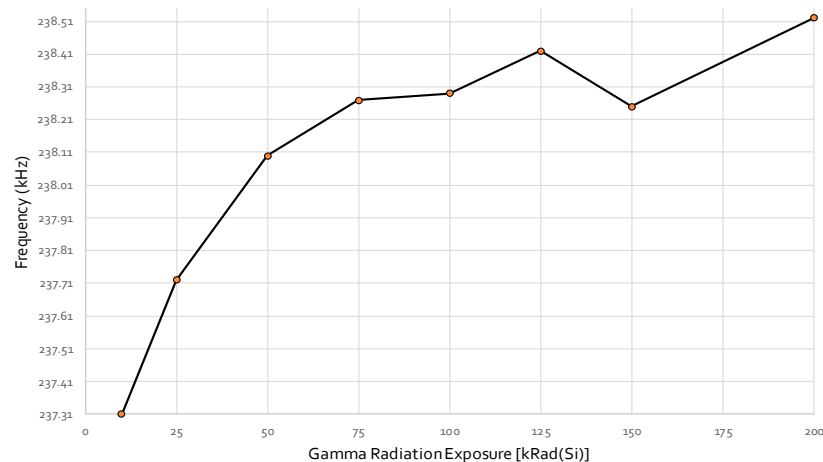
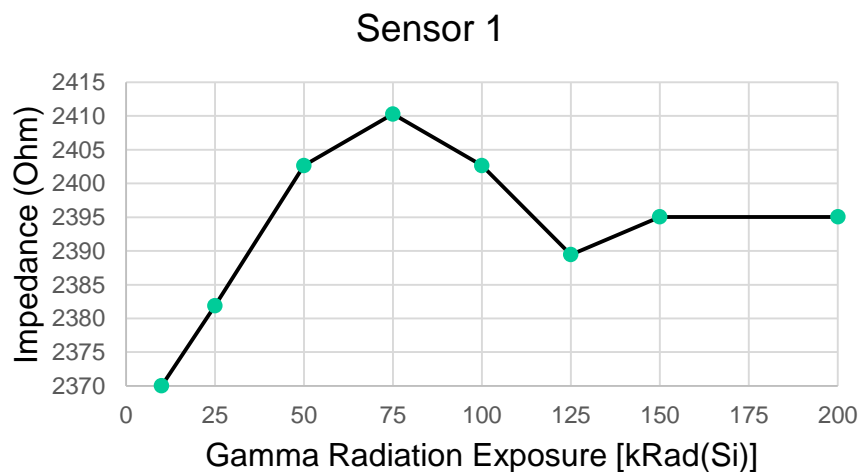
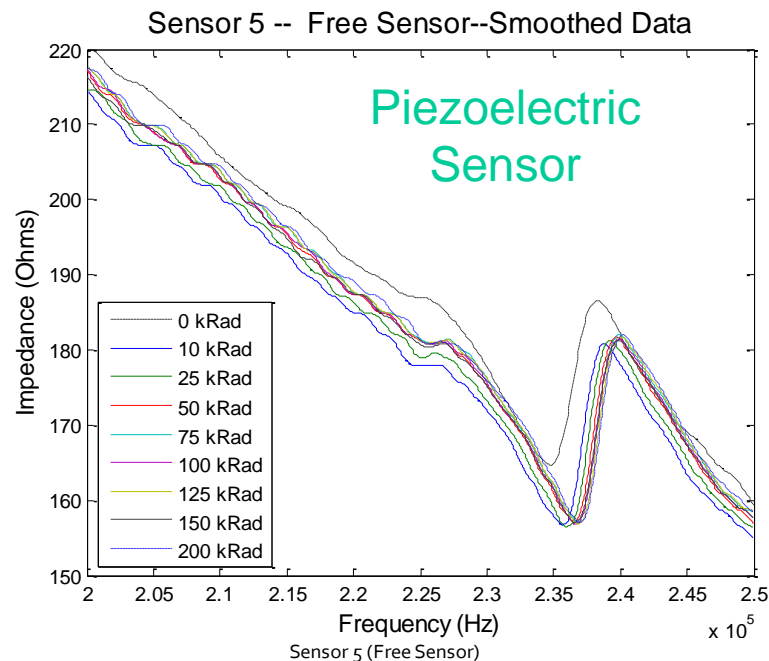
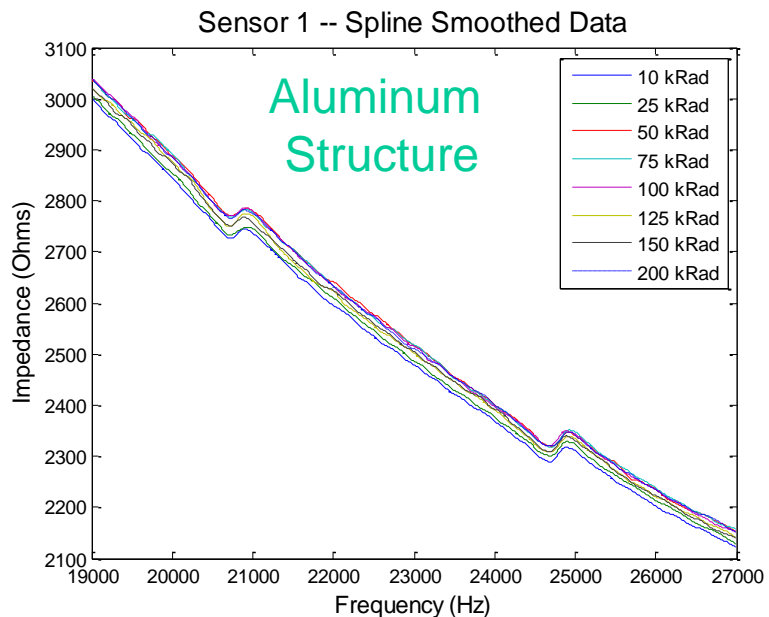
# White Sands Gamma Facility Test



# Irradiation Plan for LEO Simulations

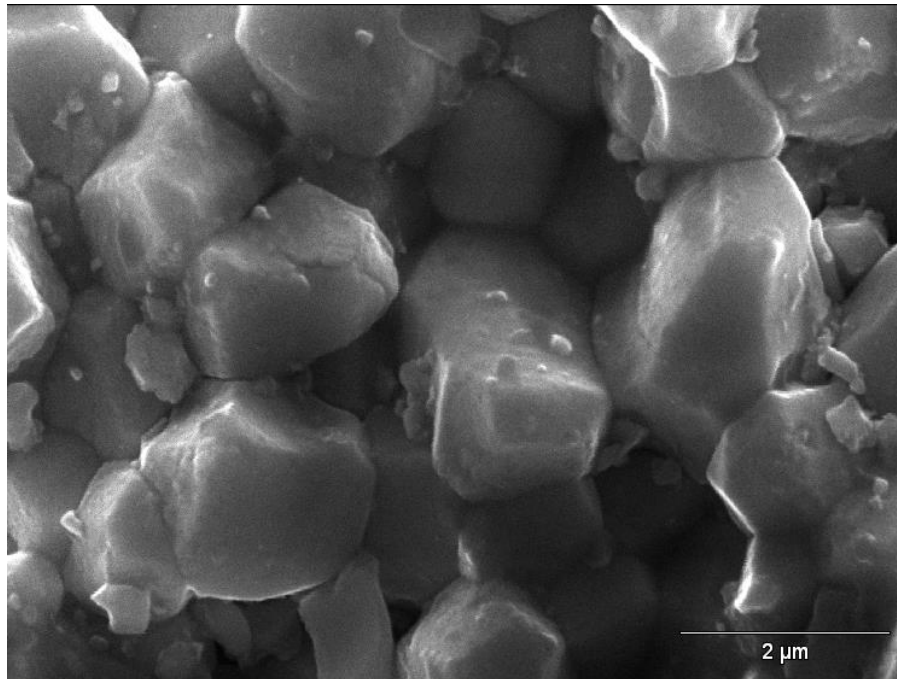
	Target Dose [Rad(Si)]	Target Exposure [R]	Actual Exposure	Estimated Dose [Rad(Si)]
	10,000	11,547	11,494	9,954
	25,000	28,868	16,119	13,959
	50,000	57,737	31,110	26,941
3 month	<b>75,000</b>	<b>86,605</b>	<b>24,560</b>	<b>21,269</b>
	100,000	115,473	33,100	28,665
6 month	<b>125,000</b>	<b>144,342</b>	<b>28,710</b>	<b>24,863</b>
	150,000	173,210	27,320	23,659
1 year	<b>200,000</b>	<b>230,947</b>	<b>62,510</b>	<b>54,134</b>

# Effect of Radiation on Impedance Signatures

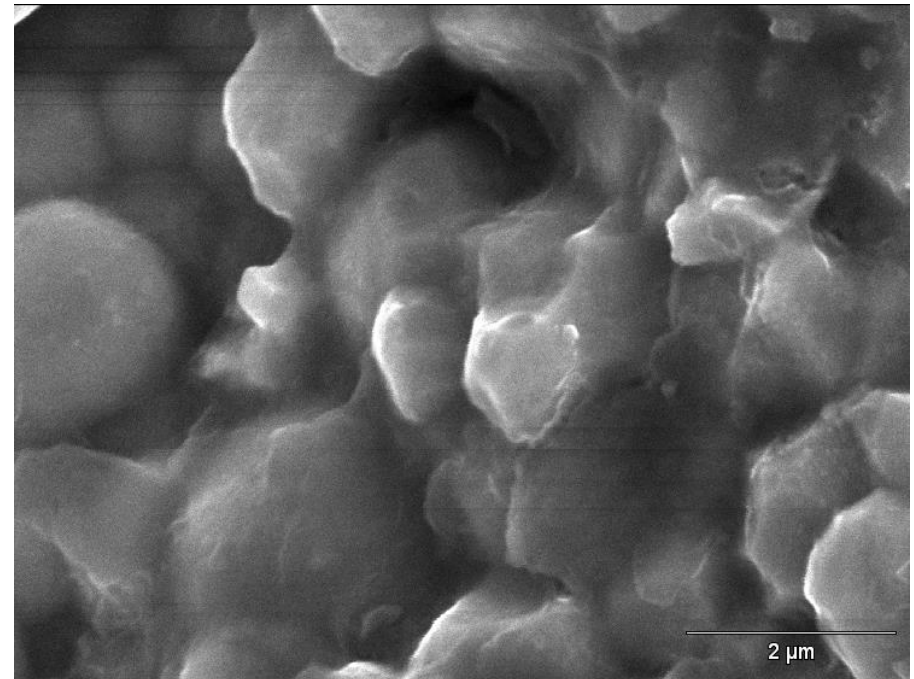


# Effect of Radiation on Piezoelectric Ceramic

Before Irradiation



After Irradiation



# Analytical Model

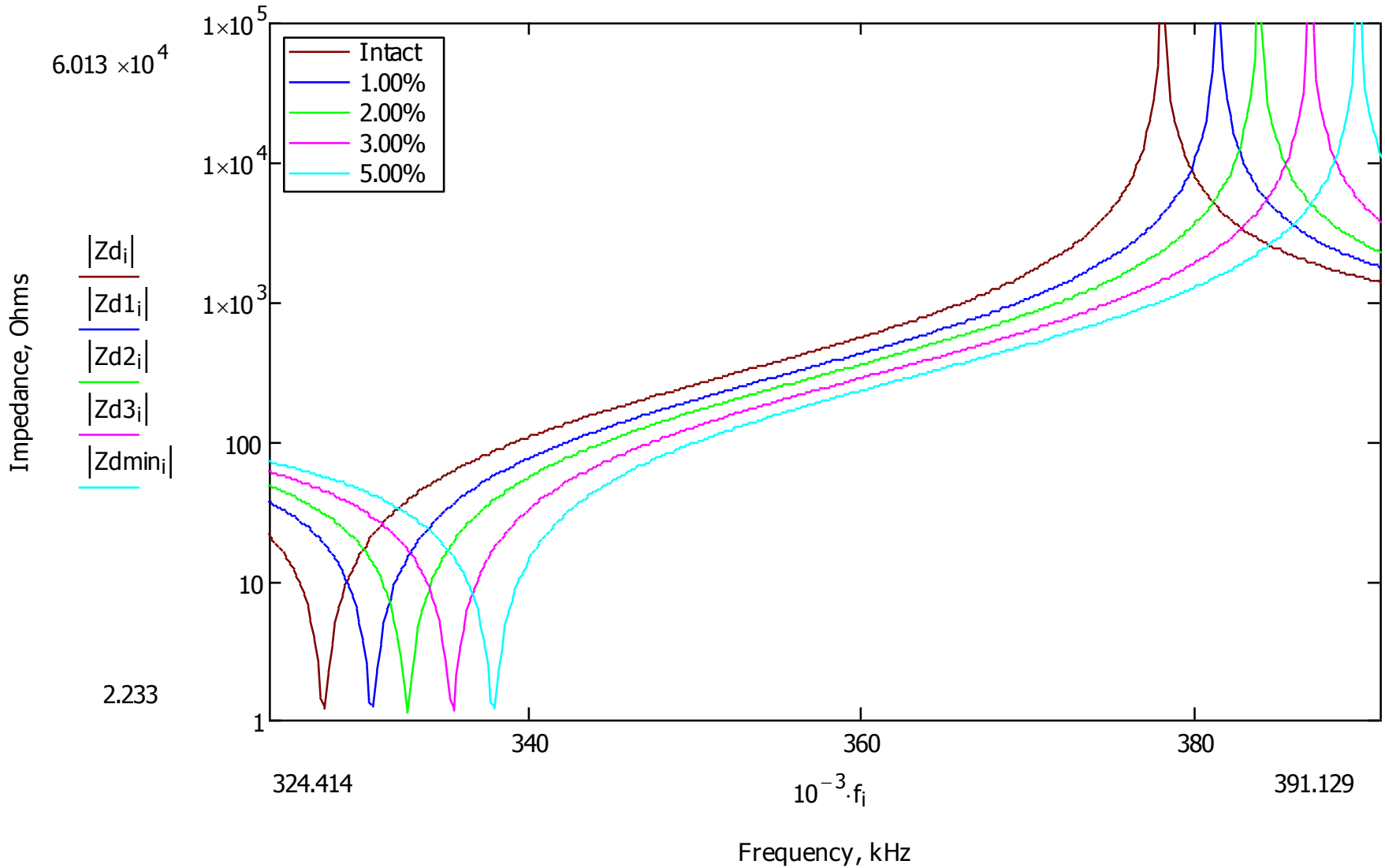
In our modeling efforts,  
we propose that exposure to gamma radiation  
causes:

- Density loss in piezoelectric ceramics
- Decreased capacitance in PZT sensors

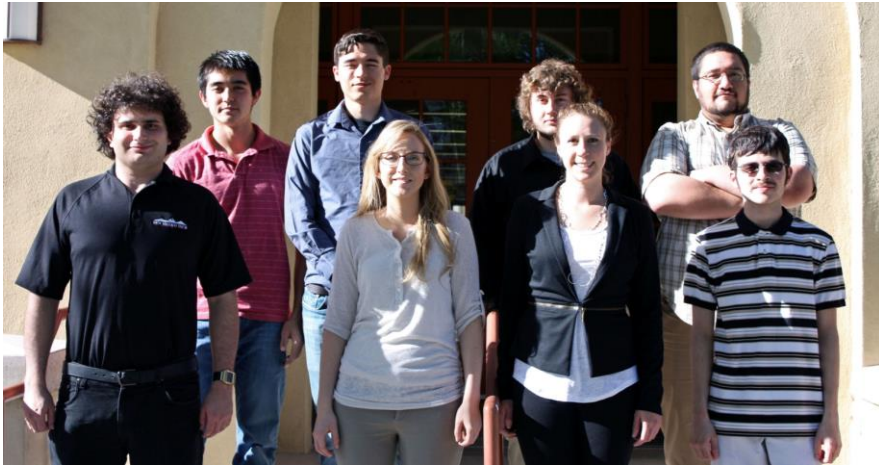
$$Y_{dmin\ i} := i \cdot \omega_i \cdot C_a \cdot \left[ 1 - (k'_p)^2 \right] \cdot \left[ 1 + \frac{(k'_p)^2}{1 - (k'_p)^2} \cdot \frac{(1 + \sigma_{PZT}) \cdot J_1(\phi_{min\ i})}{\phi_{min\ i} \cdot J_0(\phi_{min\ i}) - (1 - \sigma_{PZT}) \cdot J_1(\phi_{min\ i})} \right]$$



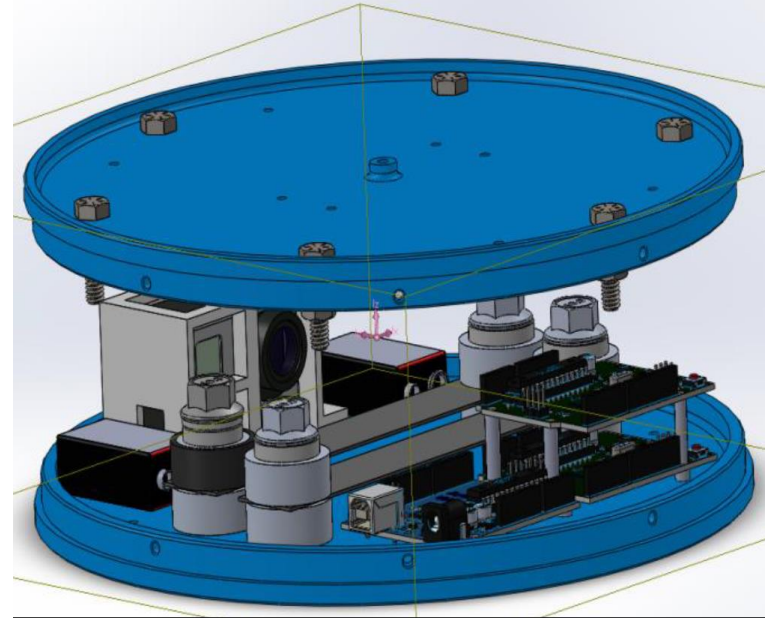
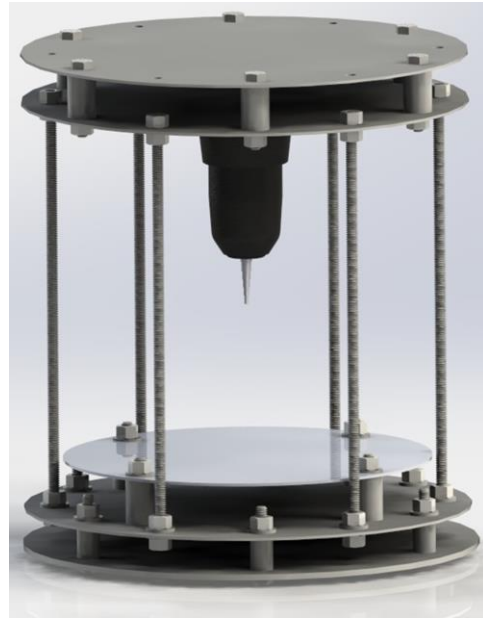
# Analytical Model



# Undergraduate SHM Team



- Develop SHM capability for space structures
- Evaluate via NASA suborbital flight opportunity (NASA USIP proposal)
- Participation in future FOP



# Acknowledgements

- Federal Aviation Administration (FAA) through Center of Excellence for Commercial Space Transportation,
  - ❑ Dr. Nickolas Demidovich
- AFRL Space Vehicles Directorate,
  - ❑ Michael R. Wilson
- White Sands Missile Range
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  - ❑ Jerame Lopez and Ricardo Ortiz, Gamma Range Facility Operators, WSMR
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  - ❑ Patricia C. Hynes
- New Mexico Institute of Mining and Technology
  - ❑ Graduate Student Association,
  - ❑ Ian Lopez-Pulliam, William Valiant and Ryan Borden, NMT Machine Shop
  - ❑ Gary Chandler, NMT Materials Department.

# Publications/Presentations

- Zagrai, A., and Demidovich, N. (2016) “Structural Health Monitoring Potential of Commercial Space Vehicles,” **invited talk** at Commercial and Government Responsive Access to Space Technology Exchange (CRASTE), June 20-23, 2016, Westminster, Colorado.
- Zagrai, A., Trujillo, B. and Demidovich, N. (2016) “Acoustic Emission during Thermal Fatigue of Aluminum Alloy,” presentation at Commercial and Government Responsive Access to Space Technology Exchange (CRASTE), June 20-23, 2016, Westminster, Colorado.
- Anderson, M., Zagrai, A., Daniel, J.D. (2016) “Potential Use of Piezoelectric Sensors For Structural Health Monitoring In Radioactive Environments”, presentation for the workshop organized by Institute of Nuclear Materials Management Technical Exchange, Taos, New Mexico, May 19, 2016.
- Trujillo, B. and Zagrai, A., (2016) “Embedded and Conventional Ultrasonic Sensors for Monitoring Acoustic Emission During Thermal Fatigue,” SPIE's 23<sup>rd</sup> Annual International Symposium on Smart Structures and Materials + Nondestructive Evaluation and Health Monitoring, 20-24 March 2016, San Diego, CA, v 9805, paper 98051K.