COE CST Fourth Annual Technical Meeting:

Space Environment MMOD Modeling and Prediction

Alan Li and Sigrid Close

October 29-30, 2014 Washington, DC

#### **Overview**

- Team Members
- Task Description
- Goals
- Results
- Conclusions and Future Work



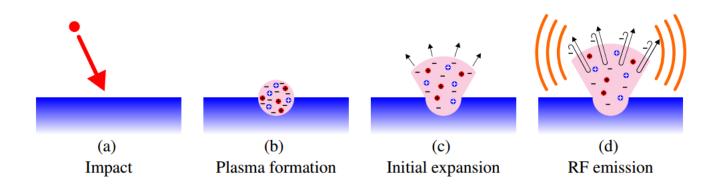
### **Team Members**

- Sigrid Close, Stanford University (PI)
- Alan Li, Stanford University (graduate student)
- Steven Pifko, Ryan Volz and Jonathan Yee, Stanford University (graduate students supported by NSF)



### **Purpose of Task**

- Spacecraft are routinely impacted by space debris and natural impactors
  - Mechanical damage: "well-known", larger (> 120 microns), rare
  - Electrical damage: "unknown", smaller/fast, more numerous



 Goal: Characterize impactor population and provide predictive threat assessment

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## Impactors

#### Meteoroids

- Speeds
  - 11 to 72.8 km/s (interplanetary)
  - 30-60 km/s (average)
- Densities
  - $\leq 1 \text{ g/cm}^3$  (icy) or > 1 g/cm<sup>3</sup> (rocky/stony)
- Sizes
  - < 0.3 m (meteoroid)
  - < 62 µm (dust)



- Space Debris
  - Speeds in LEO
    - < 12 km/s
    - 7-10 km/s (average)
  - Densities
    - > 2 g/cm<sup>3</sup>
  - Sizes
    - < 10 cm (small)

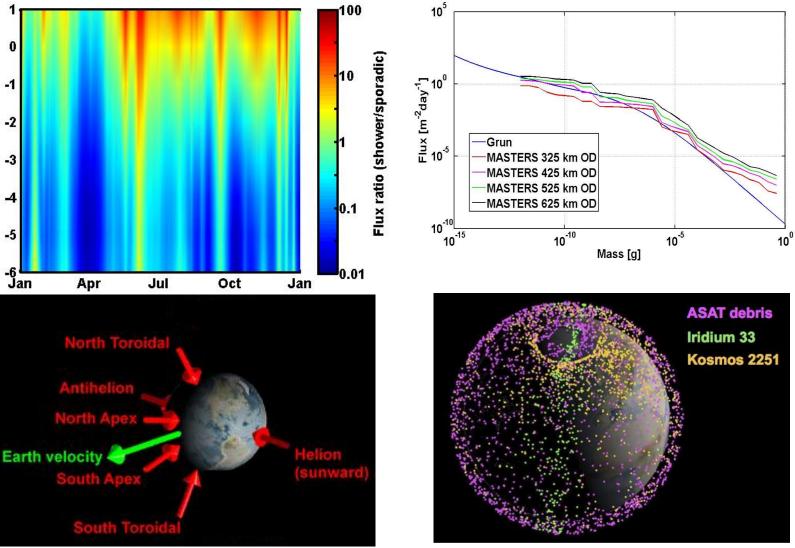






Log<sub>10</sub>(Mass) [g]

Meteoroids and Debris



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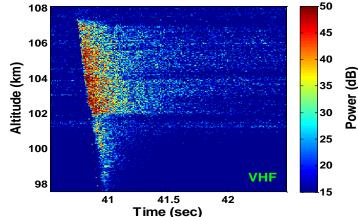
# Methodology: Meteoroids

#### Atmospheric Plasma

- Data: ground-based radar
- Models: Particle-In-Cell (PIC) for plasma
  development, Finite Difference Time Domain <sup>102</sup>/<sub>2</sub> <sup>104</sup>
  (FDTD) for EM interaction with plasma
- *Deliverables*: energy flux, mass, bulk density, orbit prediction

#### Impact Plasma

- Data: ground-based accelerators
- *Models*: Computational Fluid Dynamics (CFD) for initial conditions, PIC for plasma development and RF emission
- *Deliverables*: plasma composition, temperature, RF spectra







## **Atmospheric Data: Meteors**

#### Radars

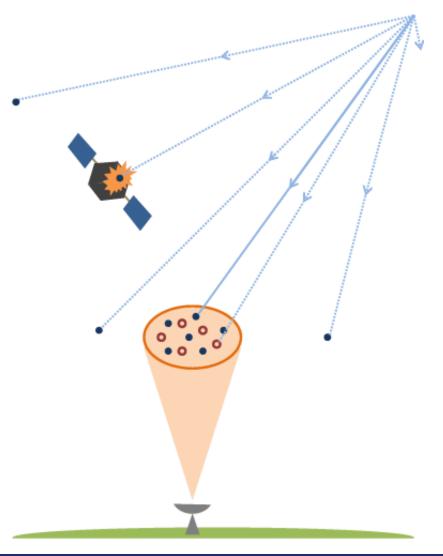
- **D** ALTAIR
- Arecibo Observatory
- I MIT Millstone
- D MU







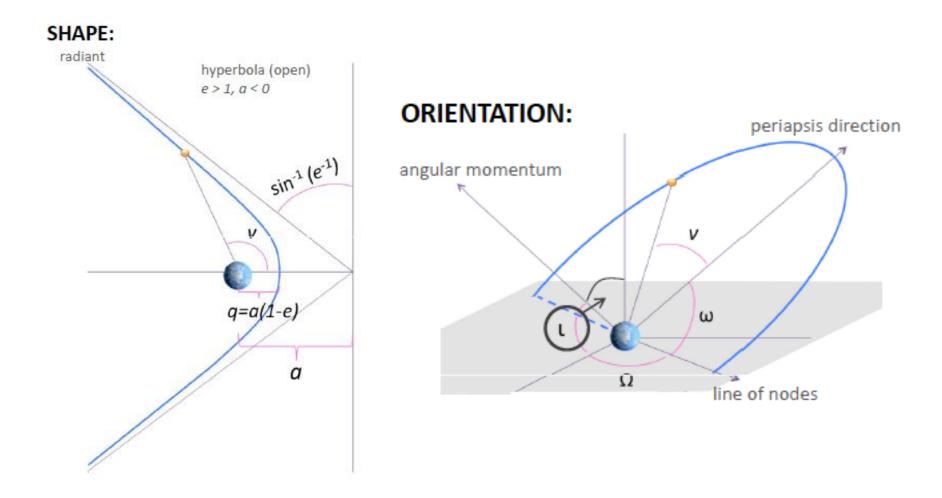




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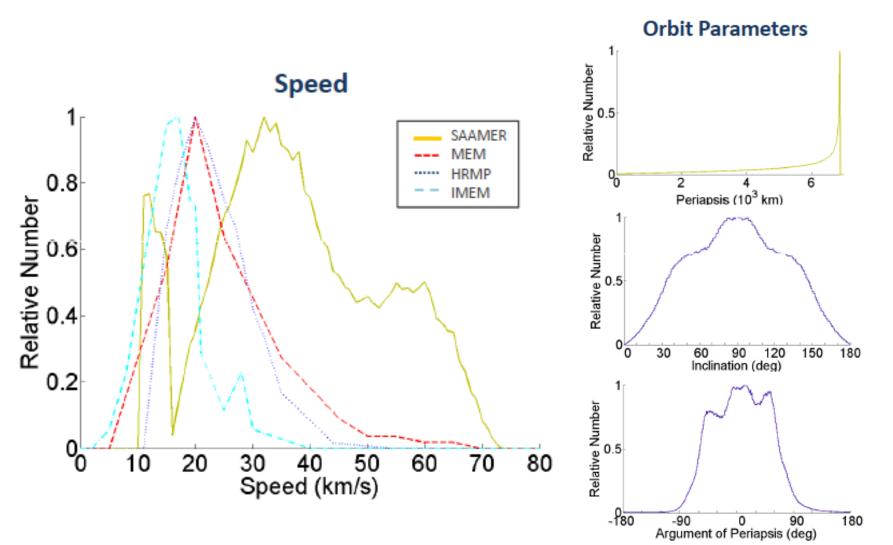
## **Orbit Determination**



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## Meteoroids at LEO



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# Methodology: Debris

#### Remote Sensing

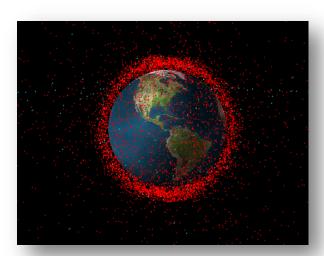
- Data: ground-based radar
- Models: ORDEM for environment, LEGEND and MASTERS for collision and propagation

#### • In Situ

- Data: CubeSats

#### Impact Experiments

- Data: future light-gas gun tests
- *Models*: Computational Fluid Dynamics (CFD) for initial conditions, PIC for plasma development and RF emission



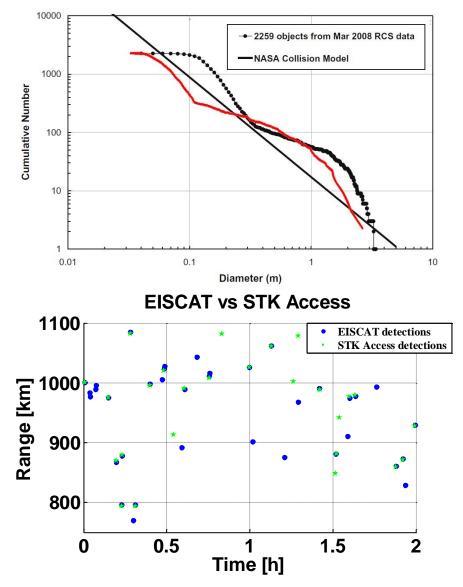




## Debris

- Characterization of debris field during impact
- EISCAT Radar
- Minimum eccentricity solution gives inclination bands
- Cumulative number of debris detected of smaller size than reported by NASA
- Correlation with Spacetrack to see untracked debris
  - 42.4% confirmed detections from ASAT test

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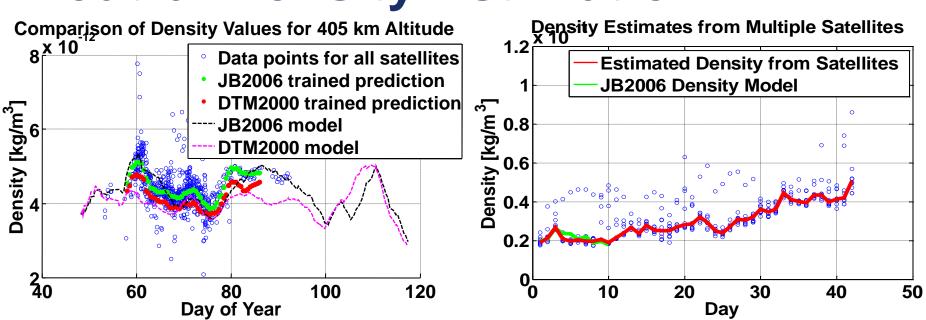


## In Situ Data and Models

- Advent of Cubesats
  - Many Cubesats in orbit, now starting in constellations
  - Known mass & area
  - Low Earth Orbit
  - Commercial Space Flight likely to remain in this region
- Drag in LEO is the predominant force other than gravity
- Models:
  - Jacchia-Bowman (JB)
  - Drag Temperature Model (DTM)
  - High Accuracy Satellite Drag Model (HASDM)
- Issues:
  - Most models have error of 10%-20% depending on solar conditions
  - Specialized drag satellites required



# **Neutral Density Estimation**



- Known minimum ballistic coefficient
- Likelihood of satellite frontal area given ADACS and decay measurements
- Calculate most likely neutral density observed with confidence intervals



## **Conclusions and Future Work**

#### Meteoroids

- Bulk density determination
- FDTD scattering
- Effect of charging on electrical failure mechanism

#### • Debris

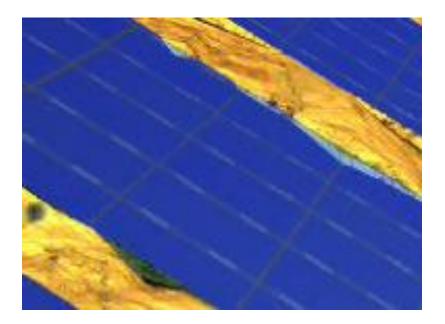
- Filtering methods for larger constellation of satellites
- Propagation of debris using near real time density data

Initial Threat Assessment Model



## **Thank You!**

- Alan Li (alanli@stanford.edu)
- Sigrid Close (sigridc@stanford.edu)



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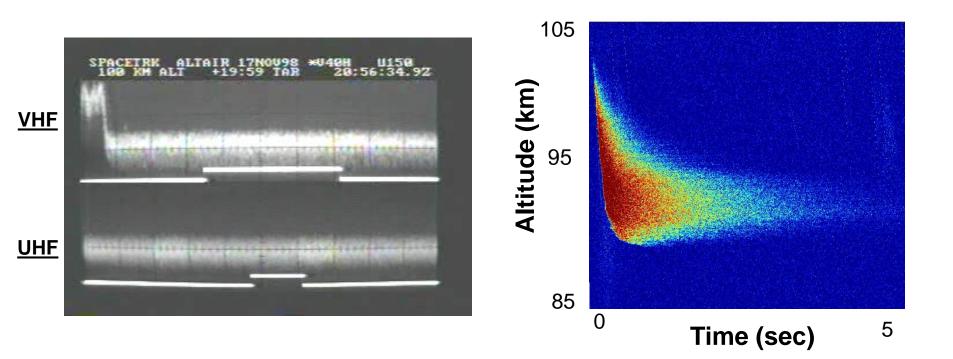


### Backup

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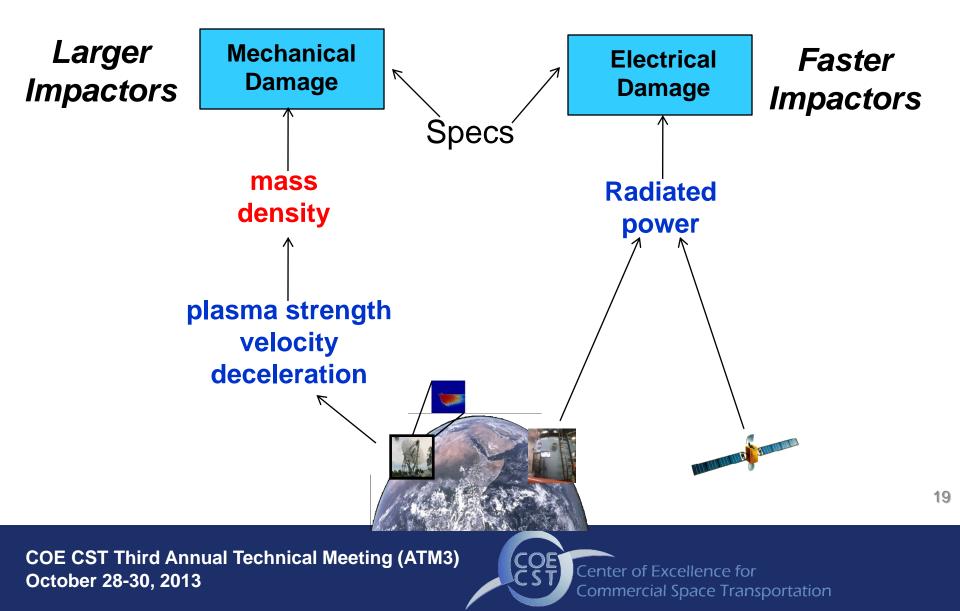
#### **ALTAIR Radar Data**



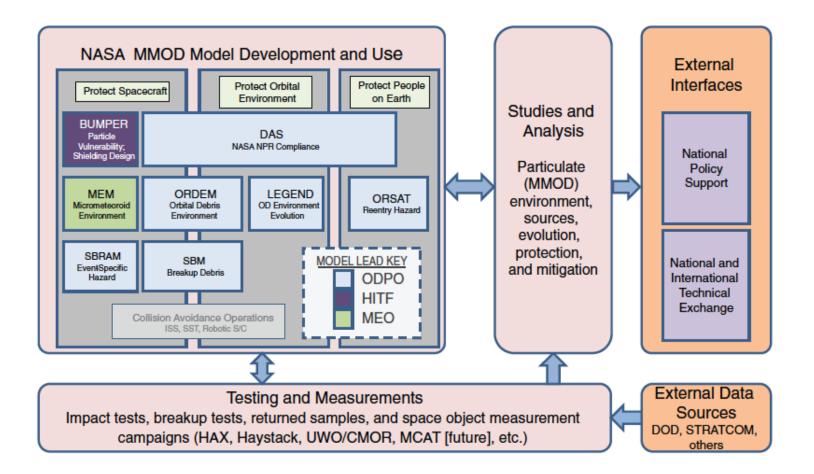
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### **Mechanical and Electrical Damage**



## **NASA Approach**



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### **Results or Schedule/Milestones**

1-2 slides

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#### **Next Steps**

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### **Contact Information**

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